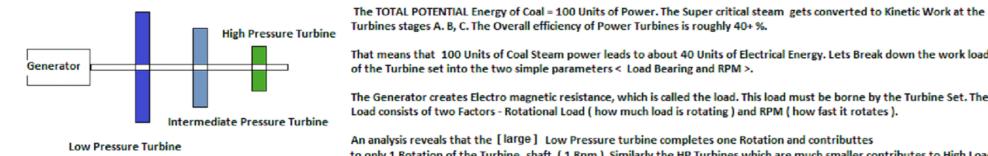
STEX ADVANCED DESIGN BUREAU - ATTITUDES IN ADVANCED ENGINEERING

POWER SYSTEMS DESIGN APPROACH USING VOLUME

ANALYSIS Alternative Turbine Blade set Design By Volume/Pressure Analysis

Conventional Power system Design upto - Upto year 2015 - 2017 ERA [A comparison]



Turbines stages A. B, C. The Overall efficiency of Power Turbines is roughly 40+ %. That means that 100 Units of Coal Steam power leads to about 40 Units of Electrical Energy. Lets Break down the work load

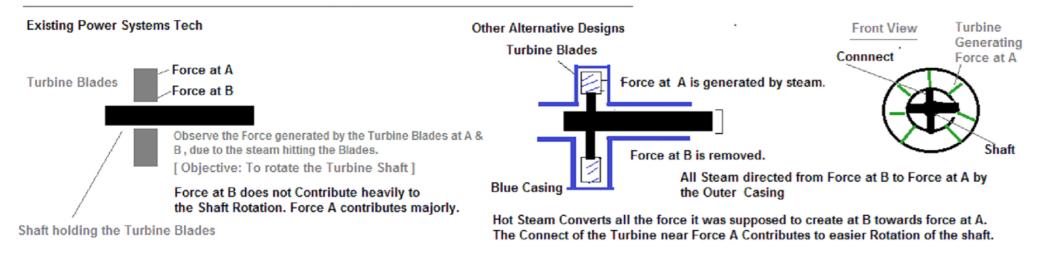
of the Turbine set into the two simple parameters < Load Bearing and RPM >.

The Generator creates Electro magnetic resistance, which is called the load. This load must be borne by the Turbine Set. The Load consists of two Factors - Rotational Load (how much load is rotating) and RPM (how fast it rotates).

An analysis reveals that the [large] Low Pressure turbine completes one Rotation and contributtes to only 1 Rotation of the Turbine shaft. (1 Rpm). Similarly the HP Turbines which are much smaller contributes to High Load (but With inefficiency, because of the Force distance - E.g. Force B as shown Below).

In the proposed next generation Systems - The HP Blades are made larger, contributing to Higher Load bearing Capacity, and the LP Turbines are made smaller contributing to RPM of the power system. The LP Small Turbines are enclosed with low volume casings to achive maximum conversion of Potential Energy to Kinetic Energy.

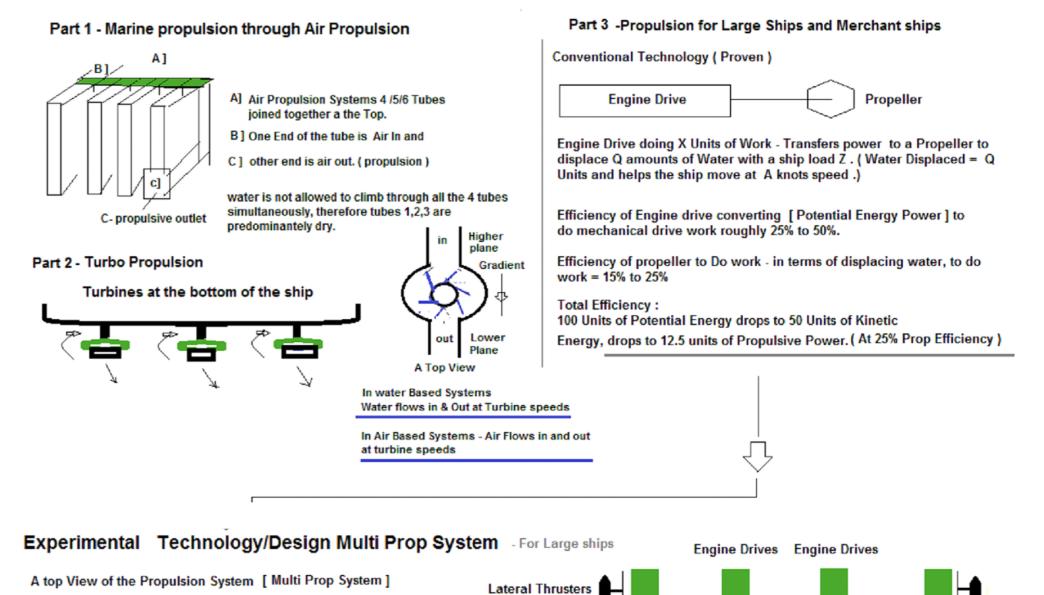
Study of Forces Contributing to rotation of the Turbine Shafts



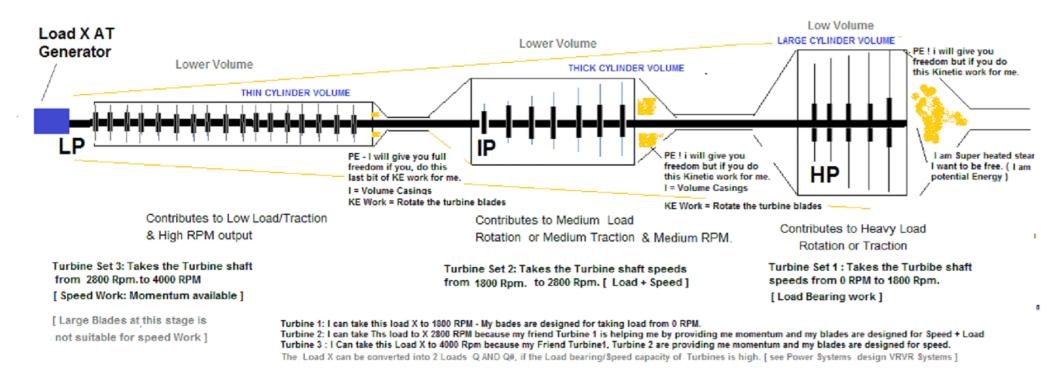
The Proposed Solution or an Alternative design leads us to building Larger HP Turbines, to contribute mainly to the LOAD/TRACTION and lesser to the RPM. Similarly the LP blades are made smaller to contribute to RPM. (1 RPM of low Pressure blades contributes to 1 Turbine Shaft RPM.)

ADVANCED MARINE PROPULSION SYSTEMS

Experimental Engg – Design Directions - IN Marine Propulsion



Efficiency of Marine propulsion could possibly be increased by Adding Multiple propellers on the same prop shafts.



Symmetric Dual side Ignition Engine [SDSI Engine]

Multi Propellers Total Efficiency : 100 Units of PE converts to 50 Units of KE, drops to (3 * 12.5 Units) of Propulsive power. = 30 TO 37.5 Units of Propulsive Power.



The goal is to Increase Propulsive Efficiency through Multi Propeller Systems.

The Load Bearing capacity or propuulsive Speed of Marine systems could be increased by this technique/design.

Most Engine drives operate at a Range of Load or a Range of Work Output : E.g. 40 Units of work output to 100 Units of Work Out put. Most Marine propulsion propeller systems are Fixed - So they operate with a Fixed average Efficiency for the Entire Engine workload range. (Some propellers are designed to give maximum efficiency at 70 Units Engine Load. They may not be very efficient at 30 Units load or 100 Units load. which is mostly a design to balance speed vs load bearing capacity. A 3 prop System combined may cover the 30 Units to 100 Units Range more easily.)

Finally the Propeller blade angles from prop 1 to prop 3 may vary - 1 works well at high Load bearing capacity (heavy lifting at low Rpm) and prop 3 works well at high speed, prop 2 works well at mid weight / mid speed conditions.

Analogy with Oars - 1 Person Rowing a Boat

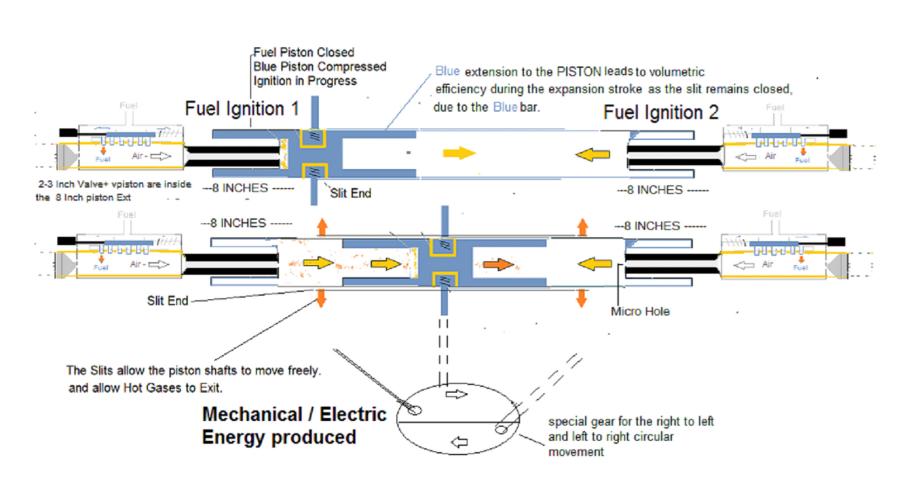
[1 person with 2 oars, 1 person with 4 oars, 1 Person with 6 oars]. 1 person with 1 oar may not be very efficient. 1 Person with 8 Oars may be too tough.

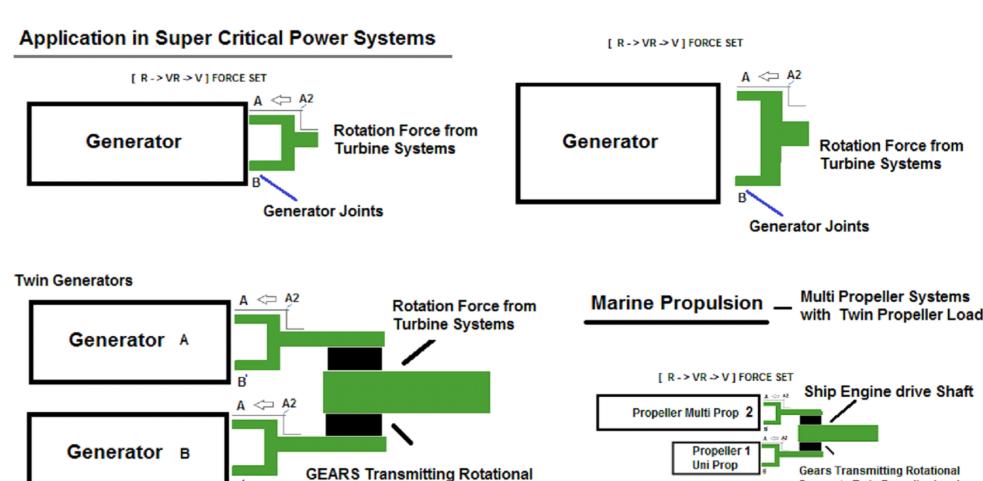
In a Ship - Under Constant Load - 1 Rpm of a Single prop system may be equivalent of 0.6 to 0.7 Rpm of a 3 Prop Systems. Total Volume of water displaced could be $3 \times .65 (1.95)$ compared to $1 \times 1 (1)$.

SYMMETRIC DUAL SIDE IGNITION -

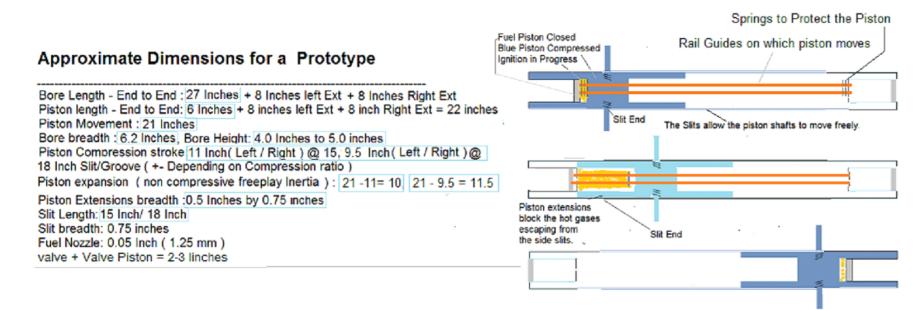
ENGINES

VRVR TECHNIQUE - APPLICATIONS IN POWER SYSTEMS & MARINE PROPULSION





Compression & Expansion Strokes With Dimensions



Forces to the Twin Generator Rods.

Micro Picture: Force Balancing Macro Picture:Load Balancing

Micro Picture: Force Balancing Macro Picture:Load Balancing

Forces to Twin Propeller Load

Pure Rotational forces are not very Efficient in Transmission of Forces. [R -> R] is not as efficient as [R -> VR -> VR -> V] Sequence of Force Transmission.

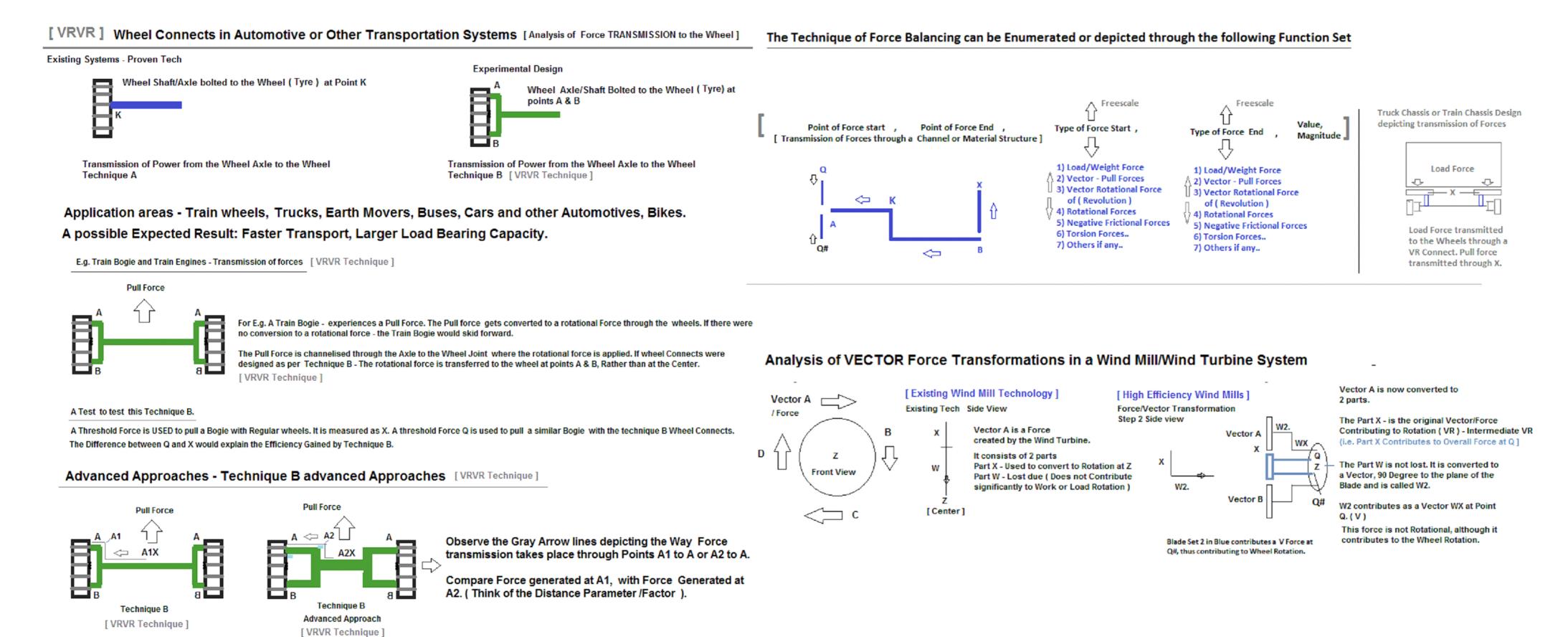
The Next Question ? Can The Gears be designed to Increase or Decrease RPM - Rotations per Minute on the Sub shafts or, Maintain Main Shaft RPM. Higher RPM or Lower RPM in the sub shafts, Depends on the LOAD at the Generator A & Generator B or Propeller 1 or Propeller 2.

[Generator Load = Elecromagnetic Resistance load] [Propeller Load - Water Displacement Volue and speed of Displacement]

THE CONCEPT OF Force Transmission, [Vectored Rotation] can be used in Fans, Motor Pumps, Fan Blowers, Rotary Systems, Helicopter blade Systems, Ground Drilling Machines, Screw Drills & Other Drilling Machines. Formula 1 Cars too !

STEX ADVANCED DESIGN BUREAU - ADVANCED ENGINERING PART - II

VECTORED ROTATION] VRVR TECHNIQUES - USED IN WHEEL DESIGN [FOR TRAINS - TRUCKS - AUTOMOTIVES]



GEAR LOAD BALANCING FOR - HIGH EFFICIENCY POWER DELIVERY

4 Channel Power Engineering design version 2 Transmission [Gear Summation] Two Hemispheres - leading to a wheel rotation Bevel+spur Fixed gears used for Gear[A1, B1 C1, D1 (Latitude, longitude styled grooves for gear engagement) gear summation fitted to grooved axles eg. 2000 Engine rpm transforms to Gear A1+ B1+ C1 + D1 + E1 = 8000 Rpm (effective 5000- 6000 Rpm) = Summation of Work Done by all gears together in serial. Forward gears Other gear Reductions n dear engagemen or Escalations axle A Reverse gears for reversing On gear engagement Note : ED Ver 1 :has fixed gears and Triple New Concept AUTOMOTIVE TECHNOLOGY Ver 2 : has gears that engage. 1) Hemisphere Gears - for Wheel Rotation Gea 2) Serial Helical Gears, leads to Gear summation (Rather than Disjoint gears) 3) One Axle for forward and second axle for reverse.

VR VR - TURBINE INNOVATIONS FOR AIRCRAFT JET ENGINES

A brief Outline - JET Turbo Fan Engine Propulsion - Evolutionary Advancements

Alternative Design Directions Turbo Fan Engines : 2016 - 2017 Era Deflects Air to Section A. B - Section A Section Blades Have higher Not very Effcient in Providing a Angle of Attack to the Air Flow force to the Turbine shaft. **Designed to Allow Higher** Volume of Air to Flow in through Section A. [Due to VRVR Principle] Turbine Rotational Speeds are increased - Increasing Air compression. Higher Compression leads to Increased Combustion and Higher Thrust [Note: Orange & Black, both the Turbine Shaft Speeds are increased] Blue Blades facilitate Vectored Rotation [Revolution Forces] i.e Revolving the turbine shaft from a distance , rather than rotating the Turbine shaft at the Root. Pic 4 The Gray Traiangles deflect air to the outer section of the Blades Pic 1 Pic 3 2016 Era Designs Orange is the Turbine blade creating a vector Channel X. Pic 4 Observe the Compression Blades Q and Exhaust Baldes K (Though X is a Short Vector). Observe the turbine Forces X is a Long vector Channel allowing Transmission of Forces to W. acting from a Distance, from the ROOT A. Pic 3 Also note the Vectored Rotation [Revolution] type Shaft Design. This is more efficient than - a Turbine Blade placed directly on the W facilitates Vectored Rotation [Revolution] Root as shown in Pic 2 [The Vector Channel X is to be of the Right Length and Strength] CONVENTIONAL SYSTEM - Force Conversions a Picture. Force Transmission Analysis of Existing Turbo Engines. Blue The Force generated by the K - Exhaust Turbines is transmitted to the Shaft a Rotational Force [R Force]. his R force Reaches the Q of the shaft. Then This R force drives the Q - Compression Turbines. Therefore the Pic 4 has a design that the K turbines create a VR force instead of the R force. The VR forces travel through W to the Q Blades and the VR force from W drives the Q Blades. [Leads to Higher Efficiency]

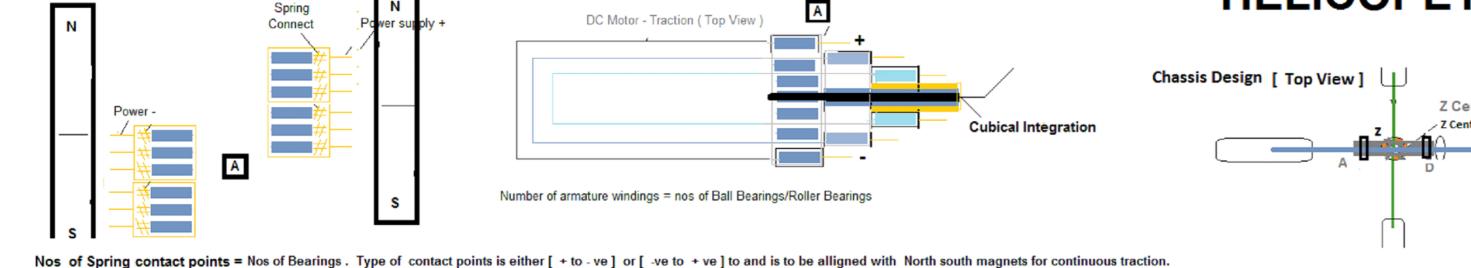
Conversion of X.X1.X2.X3.X4.X5 Forces to R

DC MOTORS FOR TRACTION

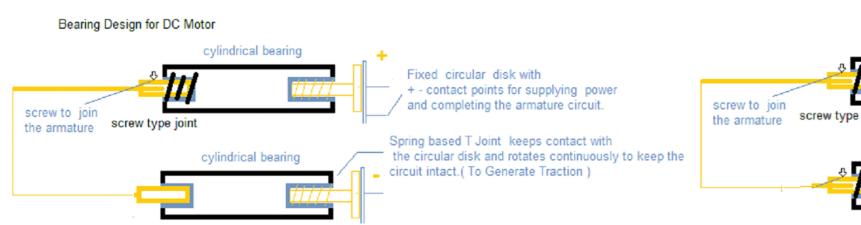
DC Motor for Traction

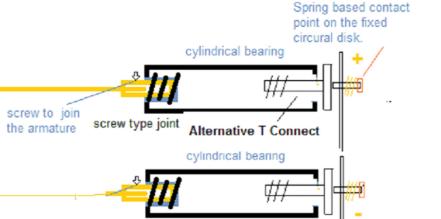
DC Motor - Traction (Top View)

VRVR TECHNIQUE - APPLICATIONS IN HELICOPETER SYSTEMS



circular square magnets are alligned in away to allow armature to gain traction for the N/2 spring contact Points with current flowing from+ To - ve. Another N/2 Contact Points have current flowing from -ve to +Ve. This Helps in creating + to - ve traction in Top Hemisphere and - ve to +ve traction in the bottom hemisphere.





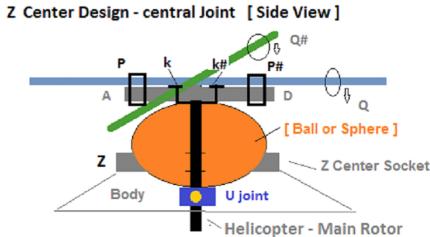
Z Center Socket Z Center - connected to the Main Rotor Shaft. For Large Angle Forward and Backward movement of the Helicopter Blades and inturn the helicopter. \nearrow

[High Angle of Attack systems]

p p# - Are Two Clamp Points where the Blue Tube is Clampped to the Gray Chassis Bar.

The P, P# Clamps allow the Blue Tube to rotate Inside.

K, K#- NUT Bolt system to fasten the Main Rotor to The Gray Chassis.



Large Orange Ball placed on a smaller Diameter Socket. Main rotor shaft - Held by a Screw/Nut joint on TOP.

When Blue Tube Rotates I.e Q, the Ball joint moves Forward/Backward. When the Green Tube Rotates, i.e Q#, the orange ball Joint moves left/Right.

U Joint to Allow - Spherical Rotation AD chassis is always at a 50 -90 Degree angle to the Main Rotor Shaft.